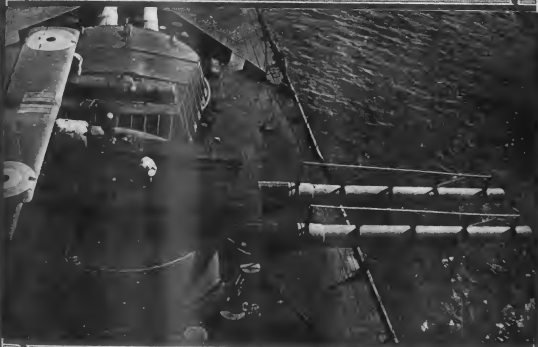


JANUARY 15, 1919

PRICE 25 CENTS

AVIATION AND AERONAUTICAL ENGINEERING



Airplane Launching Platform of a British Battleship
(U) International Film Service

VOLUME V
Number 12

SPECIAL FEATURES

PROPERTIES OF THE LIBERTY AERONAUTIC FUEL
NATIONAL ADVISORY COMMITTEE REPORT
USE OF AIRPLANES IN FOREST PATROL WORK
BRITISH AIRSHIP DEVELOPMENT AND OPERATIONS
DESCRIPTION OF THE LOENING MONOPLANE

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De Amsterdamse Handelsmaatschappij heeft een bericht ontvangen van de vliegtuig-afdeling van de Nederlandse luchtmacht, dat een vliegtuig van de Nederlandse luchtmacht op 14 januari 1919 is afgevoerd naar de Verenigde Staten. Het vliegtuig is een Martin Night Bomber, een viermotorige bommenwerper, die is afgevoerd naar de Verenigde Staten door de Nederlandse luchtmacht. Het vliegtuig is afgevoerd naar de Verenigde Staten door de Nederlandse luchtmacht. Het vliegtuig is afgevoerd naar de Verenigde Staten door de Nederlandse luchtmacht.



The ship

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Two persons who visited the machine of the machine. The machine was consequently have every an idea in the machine's machine. (Holland, Amsterdam, Holland)

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JANUARY 15, 1919

AVIATION AND AERONAUTICAL ENGINEERING

VOL. V. NO. 12

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1919?

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Vol. V

January 15, 1919

No. 12

THE annual banquet of the Manufacturers' Aircraft Association, which was this year a particularly representative gathering of men directly or indirectly concerned with aeronautics, emphasized principally two points.

The first of these was the magnitude of the industrial firms represented, all of which have since the war been engaged in the pursuit of some phase of aerial navigation. Before 1917 the aircraft industry struggled along year in and year out with insignificant government encouragement, and produced, as a consequence, but limited quantities of airplanes and engines. But with the production program of 1918, based on demands made by the war, many of the larger manufacturing companies have become extensively engaged in airplane production, and this activity has naturally led them to consider the post-war problems of aeronautics. Judging by various expressions of opinion, these manufacturers are as a rule far-sighted enough to grasp the great possibilities aircraft afford for public transportation and are making, or have already perfected, post-war production plans in accordance. The vast amount of aircraft production machinery which has been accumulated during the war as well as the engineering talent that has been developed in the pursuit of military requirements is therefore likely to be used in the future for the needs of commercial aeronautics, instead of going to waste through lack of proper comprehension of the problem.

The second note which predominated in the conversations at the tables and in the addresses made was that, while the aircraft manufacturers and engineers are determined to develop the field of commercial aeronautics regardless of prescriptive reports from official quarters, they fully realize the necessity of effecting a modification of certain basic conceptions of airplane design, so that aircraft may furnish the maximum of all-round efficiency in the pursuit of peace-time objects.

For the past ten years every effort in the matter of airplane design has had as its background the development of machines endowed with ever-increasing speed, climb, and maneuvering ability. This tendency has obviously necessitated a good many compromises in terms of maximum safety, such as comparatively high landing speeds, high wing and power loading, low factors of safety, etc. While war requirements fully justified such a course, there is no gainsaying that the demands of peace-time, that is civil, aeronautics will needs bring about a considerable change in these engineering conceptions.

Reliability must become the first and principal watch-word of airplane design and construction. Reliability

with respect to the materials used, reliability in the matter of assembly work and inspection, and what should perhaps provide these two requirements, reliability in actual airplane conception and design—in brief, all round reliability, this should be the constant endeavor of all those concerned with the manufacture of aircraft.

Commandable progress has already been made in this respect on comparison of machines turned out today with those produced a few years ago. However, the stress of war has often retarded the adoption of construction standards which were the results of hasty production methods, these must of course be eliminated if the airplane is to become structurally as reliable—which most emphatically is feasible today—as any other engineering work.

To make of the airplane a machine which will convince the public of its desirability for civilian pursuits the element of safety must absolutely predominate in every feature. To ensure this will mean that airplanes for civilian use will have to embody first of all much lower landing speed than was hitherto customary, this will not only increase safety on landing, but will also permit the use of much smaller aerodromes. Another problem that will require serious consideration is the reduction of wing and power loading, for such a course will increase safety in flight.

Besides these problems, which concern the element of safety pure and simple, there are some subsidiary questions which also demand solution, chief among these is the matter of comfort. Development along these lines, assisted by actual demonstration of what airplanes specially designed for the post-war demand offer to the public, will go a long way toward popularizing aerial navigation for civilian purposes.

There are some of the more substantial reasons which make it undesirable to use merely modified military airplanes for commercial aeronautics. The experience of the Post Office Department shows that superficially altered military machines are unable to give the full commercial service that may justly be expected from airplanes specially designed for this purpose. The sooner this fact is realized, the better will be the performances of commercial aeronautical enterprises.

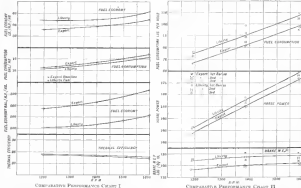
If the gathering of aircraft manufacturers and others concerned with aeronautics had accomplished nothing more than to give mutual confidence in the future of the industry as a producer of commercial vehicles, not to speak of military and naval machines, and to display the remarkably hard grasp these men have of the principal problems awaiting solution, it would have been well worth while.

Properties of the Liberty Aeronautical Fuel

The extensive investigation of gasoline which the use of two of the methods of automatic apparatus emphasizing the fuel for performance, such as accuracy, motor travel time, service endurance, endurance reserve, etc., has caused during the war years the serious problem of providing for substitute fuels to make up for the rapid depletion of the world's gasoline supply. A considerable amount of work has been conducted with this view in the United States, as a result of which an improved gasoline substitute has been developed by Capt. E. C. Tresser, of the Gas and Oil Production Division of the War Department. In accordance with the request of the General Engineering Dept. of the United States Army the Bureau of Standards has made laboratory tests of this improved Liberty fuel the report on which is printed herewith. It is to be noted that the Liberty fuel was prepared from materials and by processes, the specifications of which have not been communicated to the Bureau of Standards, but which the latter understands are in possession of the General Engineering Dept.

The tests made on the Liberty fuel were of two distinct types: (1) tests to determine those physical characteristics of the fuel which have bearing on its use in airplane engines; (2) tests to determine the comparative performance of engines when using the fuel substituted and aviation gasoline. The physical characteristics which were determined for the Liberty fuel and for the representative fuels with which it was

of Standards, from a series of careful calorimetric determinations, is a *gasoline substitute*. The conditions were made at the Bureau of Standards, in accordance with the method developed by the Bureau of Mines and described in the *Journal of Mining Technology* Vol. 16, No. 10, entitled *Motor Calorimetry* (Properties Laboratory Methods of Testing and Practical Specifications), by E. W. Dean.



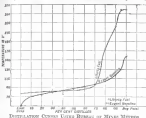
compared are, (1) the test (light) and test (dark) heating values, (2) the specific gravities, (3) the distillation temperature characteristics.

The Liberty fuel was tested in a 100 hp. type A Hispano-Suiza engine fitted with high-compression (28:1) pistons and manufactured by the Wright Martin Aircraft Corp. of New Brunswick, N. J. The Liberty fuel was compared with a representative grade of aviation gasoline, fulfilling the specification No. 3602 of the Bureau of Aeronautics for export aviation gasoline. This gasoline was distilled from Pennsylvania Crude by the Atlantic Refining Co. of Philadelphia. The heating values of the fuel was determined at the Bureau

of Standards. When a sample of the Liberty fuel was cooled to -50 deg. C. (-58 deg. Fahr.), it was found that a considerable amount of crystallization had occurred. This fact shows that the fuel as submitted cannot be cooled much below -10 deg. C. (+14 deg. Fahr.) without serious danger of stoppage in fuel lines and carburetor jets.

In the comparative engine performance test three series of observations were taken. In the first two series, five-minute runs were made at 1200, 1600 and 1600 rpm. In the third series of runs, the engine was operated at one speed, 1600 rpm, for about 1 hr., 30 min. on each fuel. During this test, oil samples were taken every twenty minutes. The analysis

of these samples was not yet complete. In all three series of runs, the engine was operated at wide-open throttle and as early as possible under constant conditions of oil temperature.



of pressure, water jacket outlet temperature, and lubricator oil inlet temperature. The engine was fitted with a General carburetor, which is so constructed that the air to fuel ratio can be varied at will from the control board. Before it was the fuel mixture was adjusted to give maximum power with the lowest fuel consumption possible. The spark advance was fixed at 20 deg. The spark plugs were cleaned and the gaps adjusted before each run.

In the second series, the revolution counter was used to check the rpm as indicated by the tachometer. Two readings were made of rpm, torque and temperature, etc., during each run at each engine speed. The revolution



counter was read each minute for a total of five minutes and showed that the rpm was fairly constant throughout each run.

The object of the third series of runs was to determine if the use of Liberty fuel would change the lubricating oil more than standard aviation gasoline.

Before starting this series of runs, the engine was warmed up and drained of all oil. Then the pump was allowed to draw about two gallons of fresh oil through the engine. After the system had drained through, 30 gal. of Mobil's Heavy Oil was put in. After the engine had been idling for a few minutes, a sample of the oil was drawn off. During the run

samples of oil were drawn off every twenty minutes until five samples were obtained.

Before starting these tests new spark plugs were put in for the run on export gasoline, and then fresh plugs were used for the run on the Liberty fuel. The plugs used in the Liberty fuel run showed a slightly greater carbon deposit than the plugs used in the run with export gasoline.

The attached curves give the revolutions reached from the observed data in simplified form.

The results of tests indicate that Liberty fuel, compared with a gasoline fulfilling the export specifications for aviation gasoline, will develop about 5 per cent greater horsepower when consuming 10 per cent greater weight of fuel per horsepower hour. The thermal efficiency of the engine when using Liberty fuel is, however, about 3 per cent greater than it is when using the export grade of gasoline.

TABLE I

Fuel	Heat Value, Btu./lb.	Specific Gravity, 60° F./60° F.	Heat, Btu./cu. ft.	Heat, Btu./cu. ft.
Liberty fuel	13,000	0.740	9,700	13,000
Export aviation	12,900	0.740	9,600	12,900

TABLE II

Fuel	Specific Gravity, 60° F./60° F.	Engine Speed, rpm	Power, hp.
Liberty fuel	0.740	1600	100
Export aviation	0.740	1600	100

TABLE III—FUEL DISTILLATION DATA

Liberty fuel	Export aviation
Initial boiling point	120° F.
10% distilled	130° F.
20% distilled	140° F.
30% distilled	150° F.
40% distilled	160° F.
50% distilled	170° F.
60% distilled	180° F.
70% distilled	190° F.
80% distilled	200° F.
90% distilled	210° F.
100% distilled	220° F.

Aeronautical Patents

ISSUED OCTOBER 12, 1938

- 2,249,440—To Theodor Schuler, Berlin, Germany. Aircraft life line.
- 2,249,441—To Theodor Schuler, Berlin, Germany. Aircraft life line.
- 2,249,442—To Theodor Schuler, Berlin, Germany. Aircraft life line.
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- 2—The design and construction for the Navy of the largest flying boat in the world, colossal crafts capable of carrying five tons useful load. It was one of these boats that recently carried fifty passengers.
- 3—The design and construction for the U. S. Navy of the fastest and most efficient Seaplane in service anywhere. This craft, which is known as the Curtiss model 11-A, with Liberty motor, made an official speed of 125 miles per hour with full military load, armament, ammunition, pilot and passenger.

The development and construction of a 12 cylinder, 600 H. P. motor of an entirely new and much lighter type, known as the Curtiss model K-12. These motors have undergone exhaustive tests and are already in production.

The development and construction of the Curtiss model K-6, a new and much lighter 6 cylinder motor. This engine develops 150 H. P. and possesses greatest endurance and reliability.

The development and construction on a large scale of the Curtiss OXK motor, and the J-N-4 training planes, which were used almost exclusively by the United States and Canada and largely in England for the training of American and British aviators. The training of over seven-tenths of the original land and marine flying pilots, most of whom entered the service and formed the nucleus of the United States Aerial Fighting Force.



The Curtiss Engineering Corporation is today the center of aeronautical activity. Its activities, instead of being decreased, will be increased by the closing of peace. Glenn H. Curtiss and his engineers have been busy in carrying forward the production of suitable commercial types. Aircraft are already available, and are as superior in design, workmanship and performance as military planes have proved themselves to be.

CURTISS ENGINEERING CORPORATION, GARDEN CITY, L. I.

British Airship Development and Operations

The British aerobus service, while generally a branch of the Royal Air Force, is in fact attached to the British Navy, since the scope of the operations is purely naval. For this reason, the work accomplished by British aerobuses during the war, as well as developments in design here, up to the summer of the present, have kept as secret as the other operations of the "naval service." Considerable interest attaches, however, to the following semi-official statement, issued by the British Air Ministry, as to efforts the first authoritative presentation of the development and operations of British aerobuses during the war. The figures given relative to man power required for operating aerobuses, casualties per flight mileage, and non-flying days, are particularly worthy of note.

The outbreak of war found the Royal Navy in possession of seven aerobuses, all of the same rigid type. On October 22, 1918, the number was 121, including rigid and non-rigid. Four of

these were over 100 ft long and had a cubic capacity of approximately 1,000,000 ft. A more recent British rigid has a capacity 50 per cent greater, while others larger still are on the stocks.

Function of Aerobuses

During the war the functions of aerobuses have been two—namely, to conduct the submarine menace and to act as scouts for the fleet. Bomb dropping made over land were not attempted by British aerobuses, and were only considered of minor importance by the Germans.

Rigid aerobuses are the best of scouts for a fleet. The Germans used their cause after the battle of Jutland to their disadvantage. It was Zeppehens which made possible the escape of the British which "refused" Scapaflow to the British Navy. In 1918, it was Zeppehens which enabled U-boats to torpedo the M.S. Torpedos and the British.

The struggle with the submarine menace entailed two sorts of work, patrolling and covering. While on patrol the aerobuses, particularly the R.N. type, were able to compare the distribution of submarine submarines, often by using their own bombs or by increasing surface craft. They were also able to locate the wreckage of a danger zone. Mines were likewise observed and destroyed.

The system of covering by aerobus proved extremely successful. No ship was ever sunk while under their observation. The work was very arduous and required constant vigilance. Aerobuses are not so independent of weather as are submarines or air craft, but it is interesting to note that in 1918 up to

the signing of the armistice there were only nine days on which no flying took place.

Man Power Required

It is a popular belief that aerobuses require an extraordinary number of man power. While the number of flying hours is not against the number of hands employed, aerobuses are shown to be more economical in man power than heavier-than-air craft, as the following figures show.

	Aerobuses	Albatrosses and Bombers
Man employed per hour flown	10	15
Hours flown per month	100	100
Manpower required per month	1,000	1,500

The following table shows the hours flown from 1915 to Oct. 31, 1918.

	Hours
Aerobuses	10,000
Albatrosses and Bombers	10,000

The total casualties to aerobuses personnel during the war were 120, of which 45 were fatalities due to flying accidents and enemy action. For every fatality suffered, 42,000 man hours were flown. In the aerobus service no one is allowed, but no losses suffered, but for losses suffered on the enemy, and for aerobuses lost on one's own side. When the potential destruction of such U-boats is considered, the services of the aerobuses in ensuring the food supply of Great Britain and ensuring troops cannot easily be overrated.



BRITISH COASTING TYPE AIRSHIP
(C) Underwood and Underwood

the rigid ones had been taken over from the Army as December 20, 1918. The first British aerobus, a "Kite" type, appeared in the air in 1907. This was of the semi-rigid type. That is to say, she had a metal framework running fore and aft along the bottom of the envelope, and in this the car was attached. She was deleted before the outbreak of war. The "Beta" class from 1910. She was a non-rigid, and the car was rigid in a profile of fabric skin and was raised the envelope. Later in the same year the "Gamma" was produced, and was the first Aerobus, designed to be fitted with a dirigible. The "Delta" class in 1912, was remarkable for the use of "dopes" on her envelope to prevent gas leakage. The "Epsilon" class in 1913, was the first of the "Zep" type, a system of attaching rigging cables to the exterior of the envelope. This was a great improvement on the rigid system. These last four ships were in use when war broke out, and the "Delta" observed for the Belgian artillery in 1915.

The very successful aerobus experiments in 1911 with a rigid, which was accidentally broken in two before she made a flight, was 2, 3, and 4 were non-rigid. Of these the 3 was the parent of the R.N. (Rohmstedt) fleet type. No 2 was in Japan. The 4, which was a rigid, was intended to be fitted with a dirigible. The 3 was a rigid, and was subsequently employed for a while in Belgium. No 4 was a rigid, and was subsequently employed for a while in Belgium. No 4 was a rigid, and was subsequently employed for a while in Belgium. No 4 was a rigid, and was subsequently employed for a while in Belgium.

During the war time rigid and non-rigid aerobuses have been constructed in Great Britain—General, Rohmstedt, Zeppelin

and built five more. Aerobuses of this type did most of the long distance patrols during the last two years of the war, and were largely employed in searching ships from beyond the British lines up the Channel. Their work has been most satisfactory, but they are now being supplanted by an improved line known as the "F" (General Staff). It is a slightly larger (211.5 ft long and greatest diameter 121 ft), and the envelope is of a better stress line shape to minimize air distribution. The first "F" appeared in January, 1919, and one more were added before Oct. 31. The "F" had a 110 ft. boat forward and a 200 ft. flat aft.

The "North Sea" type was designed as a vessel to act with the fleet or to carry out patrols of 30 ft. Its envelope has a capacity of 300,000 cu ft., and the car can carry a crew of 10. Its normal speed being 10. Since the carrying of the armament a North Sea ship has a flight of 61 ft. It is rigid, and is a vessel for a non-rigid, and is believed to have been constructed with two performance of 60 mph. Zeppehens. The length is 303 ft., greatest diameter 15 ft., and it is fitted with two 175 hp. Rapiers or two 300 hp. Fiat engines.

Rigid aerobuses

After the introduction to the first rigid aerobus as rigid was provided to the British Navy in December, 1918, so that the command was for the first two years of the war some twelve years before the construction of rigid aerobuses. By July 1, 1918, however, four British rigid were in commission. This type is composed of frames of duralumin alloy. The whole framework is encased in a fabric cover. Inside are the gas bags, which are thus protected from the action of sun and moisture. No 23, which has recently been sent

Description of the Loening Monoplane

In his extensive report on American aviation recently made to the British Air Ministry, and forwarded through the Technical Service of the Air, Col. Lord Dunsford, R. A. F., reported in detail upon the Loening monoplane as one of the most advanced and original designs that have ever been produced, and with such great promise of success that it was well worthy of the most extensive development and study. American officers of both the Army and Navy Service, and several American engineers, including Messrs. Coase, Wilbur, and

In the tests made by the manufacturer at Mineola during last summer and fall, the machine showed a high speed of 140 m.p.h., and on one occasion climbed to 24,000 ft. in 43 min. with pilot, passenger, two hours fast and considerable gun equipment, thus establishing an additional height record for monoplanes. The new monoplane of the same type tested at Dayton by the U. S. Army showed practically the same high speed, and climbed with a less load about equal to the weight of the airplane—13,000 lb. in 45 min.



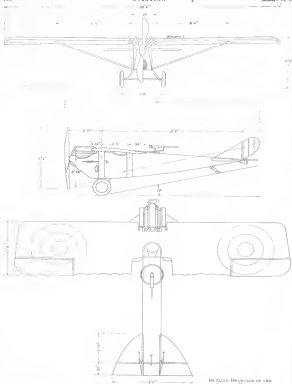
THREE-QUARTER REAR VIEW

Yough, who knew what George C. Loening had been working on for many years, were much interested and encouraging in their attitude. That when the first American monoplane was first shown by the War Department after its tests at Dayton, while surprise was shown simply by those who had so strenuously maintained in the past that American designers were incapable of producing the most advanced types of fighting airplanes by their own efforts.

Recently speaking, as to actual performance the Loening monoplane has, with the full two-seater fighter load carried by the Dunsford, Lefebvre, Boudet, etc., not only exceeded all the performances of these machines, but has with the same load equalled, if not exceeded, all of the performance of the very best European single seaters with the same engine, such as the Spad, Sopwith, Morane, etc.

The American aviation world has followed the progress of the new design with great interest, and there is a widespread general belief that the war should have ended before this machine was demonstrated, in actual service, the superiority of American high-speed fighter design. However, there are so many other uses that a machine of this type can be put to that emergency development is looked forward to and presented by Mr. Loening.

The first monoplane of this type was, as a matter of fact, ordered from Mr. Loening by the Wright-Martin Aircraft Corp. for the express purpose of serving as flying laboratory for the 300 hp. Hispano-Suiza engine of very high performance. The military advantages of the design, however, were so obvious that the machine was quickly taken over by the Air Service of the Army.



OUTLINE DRAWINGS OF THE
LOOMING MONOPLANE

During the tests conducted by both the manufacturer and the Army at Maxwell and at Dayton, a dozen or so expert military aviators repeatedly ran the machine through every conceivable maneuver, such as Mr. Looming has frequently pointed out—flurry, zigzagging and somersaults, in particular that of Capt. W. F. Jordan, Lieut. E. A. Wolfe, Capt. Cash, Major, Col. Smith-Barry, R. A. F., O. M. Bonds, Major R. W. Schofield, amongst others, has been of great value in developing the final points of the machine.

One of the most important contributions to the art of aviation

without an actual sand-test a load of 35 lb. per sq. ft. area. Throughout the machine, it is apparent that the design has been particularly studied for production on great quantities as the machine has been found to be extremely easy to build.

Military Features

The visibility afforded to the pilot is so complete that he has practically no blind spots on either side of the machine or below the wings or to either side, and in addition to that can see quite well in the front, due to the narrowness of the body



Side View

that it is made by the Looming monoplane is in the novel design of the wing structure which gives a great strength against drift currents and makes the whole machine a rigid unit, quite as strong and solid against twisting stresses as any machine. The stabilizing feature of the machine is the manner, original with Mr. Looming, in which the wings attach to the upper body longitudinally, and are braced by two cross-braces in the bottom of the body. This method of wing bracing has so much simplified the construction of the machine, that it has permitted to cut in half the weight of airplane structures heretofore used.

When it is realized that this machine weighs with the same engine as the same just over 400 lb. less in total weight than, for example, the Bristol biplane, the greatly superior performance of the Looming monoplane are readily explained.

A very interesting feature of the design proven by the tests made in the above stated on landing and ground behavior of the machine, despite the fact that the wing loading has gradually been stepped up to almost 15 lb. per sq. ft.

Construction

The wings, body, tail surfaces, landing gear, etc., are all built up of metal upper and metal fitting airplane construction. All metal fittings are of stamped sheet metal, with practically no brazed or welded parts; all joints are pin connected, particularly the joints of the main wing braces to the wings and body, which are fast to make in any direction, so that vibration will not fatigue these members. In addition to which, all parts are readily adjustable for alignment.

Safety Factor

Wind load tests that have been carried on experimentally to prove the strength of the machine show a safety factor of 14 on drift stresses, and a safety factor of 5 on lift stresses. The machine has frequently been shown there is no evidence that the wing structure is weak in any way, and wind load tests have shown that, owing to the deep chord and the reinforced manner in which the wings are braced, the usual failure of monoplane on drift stresses has been eliminated. But the wings on edge are strong enough to take a backward thrust as they are required in the weight of the entire machine with its ample factor of safety.

All tail surfaces and tail surfaces, stabilizers, rudders, etc.,

These features, as mentioned on this machine, are the results of three or four years experimenting by Mr. Looming which gives him priority over all others in the creation of this type of monoplane.

The Gun Range

The gun range is also very good, particularly as the gunner can shoot forward, the only obstruction being the arm of the observer.

The deep body offers ample room for all kinds of military equipment, engine tanks, window apparatus, cameras, etc., and in addition to that the arrangement of the seats is such as to give the occupants ample protection against the wind without interference with the view. The construction has everywhere been studied so as to give maximum strength to all of the parts that are vital due to damage from bullets. This is particularly true of the main braces supporting the wings which can be built shot away before they will lose the safety factor of a jointed.

Engine Installation

The engine installation of the machine is particularly well studied. Although the engine is left out in the open, it is found to be in a more secure position than usually, while it vents upwards the performance of the engine, and its durability, so it is not subjected under a load. The radiator is placed in a particularly advantageous location where it gets a full blast of air and in addition can be easily shifted. Practically all maintenance on the engine is accessible without taking off any rods, and the whole installation is accordingly easy to inspect and to mend. The engine can be taken out without taking off the radiator, and the latter can be taken off without removing the air-cooler. In addition to that the design is so worked out that the front bulk head, instrument bench, engine box, cowling, radiators, etc., all being one solid, made and assembled on an sub-assembly and packed as a unit, which greatly facilitates the final assembly of the machine.

Performance

In reference to the flight performance of the machine it has been found that the machine descends very slowly, gets off the ground in 4 seconds from a dead start, and that in light the machine is very easy to handle as all its controls are simple, simple, turned side, which means that the machine appears to make with the observer at a single water.

News of the Fortnight

Our Air Service in France

A brief review of operations of the American Air Service in France, up to November 15, is contained in a cable from Major General Harbord to the Director of Air Service. It is dated December 15 and below is a summary.

There were 12 operations on the front thirty-nine American Army Squadrons, distributed as follows: Twenty pursuit, one night bombardment, six day bombardment, five army observation, twelve corps observation and one night observation. Enemy planes, brought down by American flyers, included 441 captured and 554 unaccounted, making a total of 995. A total of eighty-two enemy balloons were reported as downed, of which 57½ were confirmed. On the other hand, the Air Service had only 271 planes and forty-five balloons.

Concurrent with our personnel casualties, there were 186 killed, 102 wounded, 200 missing, twenty-seven prisoners and three missing, making a total of 442.

The Air Service included in the case of advance 2363 officers, 22,253 soldiers, a total of 24,615 at the actual front. There were also 4813 officers and 36,953 soldiers in the service of supply. With the French armies there were detailed eight American Army officers, and with the British Expeditionary Force forty-two officers and 123 soldiers. The total personnel in France consisted of 6963 officers and 51,220 soldiers, a total of 58,183. Air Service mechanics regiments with the French Army included 508 officers and 4744 soldiers.

The French personnel included 1320 pilots, organized as follows: preliminary, 126; advanced, 29; pursuit, 850; observation, 149; day bombing, 77; and night bombing, 165.

Observers in training included 563 artillery, 65 day bombing and 61 night bombing, a total of 689 officers. The French personnel awaiting instruction included 134 pilots and 58 observers. Graduates included 6089 pilots, divided as follows: Preliminary, 1273; advanced, 2320; pursuit, 1382; observation, 725; day bombing, 529; and night bombing, 25.

Also a total of 3043 observers divided as follows: Pursuit, 84; artillery, 146; day bombing, 100; and night bombing, 173. A total of 159 individuals is reported as having been killed in training.

The number of planes, by type, received from all sources by the A. E. F. between Sept. 12, 1917, and Nov. 15, 1918, is as follows:

Pursuit—Army, 2227; pursuit for schools, 69.
Observation for service, 627; for schools, 664.
Day bombing for service, 422; day bombing for schools, 83.
Night reconnaissance, 31.
Other planes received included 2385 training planes, 38 experimental planes and 106 miscellaneous, making a total of 16,473.

Eight different schools under American control were established in France and designed for training 3600 officers and 11,590 men, as follows:

Pursuit—Observers: 416 officers and 3113 soldiers.
Bombard—General: 2773 officers and 6782 soldiers.
General—Pursuit—Bombardment: 159 officers and 658 soldiers.
94 days for flight—Aerial gunners: 92 officers and 1580 soldiers.
Recon—Artillery for service: 256 officers and 756 soldiers.
Compass—Artillery for service: 25 officers and 123 soldiers.
Messengers—Artillery for service: 36 officers and 110 soldiers.
Chaffin air base—Observers: 204 officers and 373 soldiers.

New Firm with a New Object

The United Aircraft Engineering Corp. has just been organized by E. G. Wells, Ray D. Conger and M. B. Stout, with offices at 161 Madison Avenue, New York, with the object to promote the development of commercial and military aeronautics, to make the adoption of the leading aeronautical engineers available in the industry in the country, and to develop aircraft engineering, the work to be international in the scope.

B. Russell Shaw Invents Vernometer

With the plans from the Airplane Engineering Division of the technical men who have been serving the Government during the war, the industry will be able to obtain detailed information of recent developments in aeronautics at first hand. Among those who have left the service to take up the work of consulting aeronautical engineer is B. Russell Shaw, who during the war was Assistant Director of Flying at McCook Field, where all American and foreign machines were



tested under his supervision. Mr. Shaw is active interest in aeronautics dates from 1909. He has built and flown a number of machines of his own design. In the early days he was in charge of the engineering department of the Wright brothers, and before entering war service was a designer for the Wright-Martin Aircraft Corp.

Mr. Shaw has recently developed an instrument called Vernometer, which indicates to the pilot at a glance the rate of ascent and descent.

The instrument is small and easily mounted on the dash. The readings are made easy by a direct and large scale. The Vernometer consists of a head in the instrument proper and a scaled-plated window line and scale mounted on the under side of the airplane body. Any small variation in the steady climb or descent of the machine is instantly shown on the instrument, which makes it possible to maintain a steady climb.

The Vernometer requires no wires or a remote actuator and is light in weight. It is started at any altitude, and the reading is not affected by opening or closing the engine throttle, as is the case with all instruments now on the market. The instrument also permits steady climb flying through a fog by indicating to the pilot the instant he has reached or passed his maximum climb.

The Vernometer is manufactured by the Aeromatic Instrument Co., 22 East 34th Street, New York.

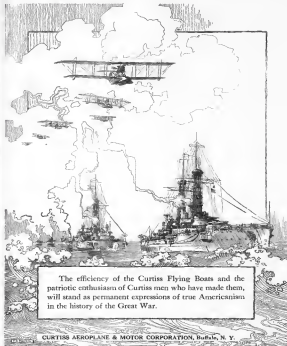
New York to Have Flying Police

New York City Police officers will begin work next Spring when the new police aviation system is put into active operation. This announcement was made December 22 by Mayor Walker L. Fawcett, president of the Board of Governors of the Aviation Section, who said that plans were complete and that the training of the police had begun already.

The school is under the supervision of Special Deputy Police Commissioner Hudson Wagoner, and is a department of the Police Department. Harry S. Dickinson is Chairman of the Police Committee and Jefferson D. Mott Thompson is in charge of the Finance Committee. The school was moved December 3 to the Arsenal Police Station, in Central Park at 34th Street, where 150 men are in training under 1 Capt. A. L. O. Finner.

Polar Exploration by Air

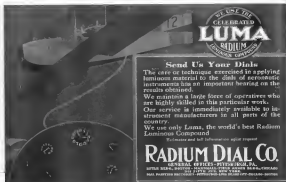
Announcement is made that plans are making for the exploration of the north Polar region next summer by means of airplanes. It is understood that Capt. Robert A. Bartlett, who commanded the Albatross in a previous expedition, is slated to head the new venture.



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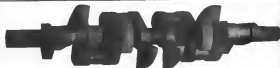
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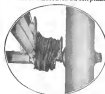
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